

AVIATION WEEK

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JUNE 19, 1950



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EX-1



B.F. Goodrich



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ONE PLACE where there was room for improvement in De-Icers was the control system. Pilots wanted closer, more complete control of the de-icing operation.

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B. F. Goodrich De-Icers with this latest type Eclipse-Pioneer plumbing system are now in use on all new 743A-type Conquestliners—as well as several other transport and military aircraft in the U. S. and Canada. This is another example of BFG research in solving icing problems. The B. F. Goodrich Co., Aeronautical Division, Akron, Ohio.

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Executive and Editorial Offices: 220 West 42nd St., New York 36, N. Y., Phone LAngston 6-0330, National Press Bldg., Washington 4, D. C., Phone National 5-1111

Domestic News Bureau: Atlanta 12, Birmingham 12, Chicago 12, 120 N. Dearborn Ave., Cleveland 15, Dallas 12, Denver 12, Detroit 12, Kansas City 12, Los Angeles 12, Miami 12, Milwaukee 12, New Orleans 12, New York 12, Philadelphia 12, St. Louis 12, San Francisco 12, Seattle 12, Washington 12, Wichita 12. Correspondents in more than 50 major cities.

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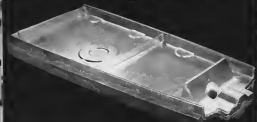
Aviation Week... June 19, 1950... Vol. 52, No. 25
PUBLISHED BY THE AVIATION WEEK COMPANY, INC., 220 WEST 42ND STREET, NEW YORK 36, N. Y.
OFFICE: 220 West 42nd St., New York 36, N. Y.
TELEPHONE: MU 2-1111
CIRCULATION: 10,000
SUBSCRIPTIONS: \$5.00 per year in advance
SINGLE COPIES: 15c
ADVERTISING: Write to the Editor, Aviation Week, 220 West 42nd St., New York 36, N. Y.
MAILING: Second-class postage paid at New York, N. Y., and at additional mailing offices.
POSTMASTER: Please send address changes to Aviation Week, 220 West 42nd St., New York 36, N. Y.

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New Foreign Planes in the News . . .



FEEDERLINE TRANSPORT

The Havilland Herald is the six-engine light twin shown above. Right side view shows resemblance to the Bristol and Dove, but a look at its underside

It is designed for economical operation on stage lengths of 500 mi. It is priced at \$99,000.



ADVANCED TRAINER

Aero Athens returns to the factory after completing a 31 month demonstration tour of India and several Middle East countries. Note unusual condition. The only photographs appeared during the tour were a group posed on the ground and a view of the tail. The trip to India was made in eight stages at an average speed of nearly 250 mph. Fuel consumption, 11,200 lb., was 177 gph. The Athens is powered by a 1800 hp. Rolls-Royce Merlin.

AVRO SHACKLETON

Long-range oceanic reconnaissance plane shown here is the Avro Shackleton. It is designed to replace the obsolete Lancaster and is being used by RAF Coastal Command. This production line model Shackleton did fly from the original model on that note and tail fin have been changed. Flying only about 1000 ft. The four Rolls-Royce Griffon four-cylinder engines are mounted in a line and are powered by a new type of variable-pitch propellers.



MACCHI MB-320

New two-engine design with an, making the pilot. Powered by 150 hp. Continental, top speed is approximately 280 mph. Construction is all-wood. Landing gear is retractable, with nose wheel folding forward and up into wing.



ITALIAN LIGHTPLANE

MB-320 is an all-wood two place personal plane by Macchi. Top speed is about 125 mph, with 85 hp. Continental C90 installation. Armed with two 500 gr. and 50 lb. engines, top speed is about 120 mph. Automatic prop is used.

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WHO'S WHERE

In the Front Office

Arthur M. N. Brown, Jr. has been named president for Edo Corp., and **William K. Ryan** has been made vp in charge of engineering. Brown was previously Edo's general sales manager. He joined the company in 1935, as 1938 became manager of the Fort Worth plant and two years later was named assistant of Ford's aircraft manufacturing division. He joined Edo in 1947. Ryan, previously director of engineering, joined the company in 1945. Brown succeeds George Pratt, who retired because of ill health last year. Ryan takes over the position vacated by R. V. Koenig, recently gone with Stevens Institute.

David D. Mason has been appointed vice-president of Lark Aircraft, Inc. He was formerly controller of Avco and has been with the production and assembly control section of Bristol Aviation.

Changes

With the **Messerschmitt-George H. Seltzer** has been named chief engineer in charge of design and engineering for the Sales Dept. **Robert E. Work** and **Edith D. Ferguson** have been appointed manager and sales manager, respectively, of Bellco. **Franklin J. Korman** is general manager. **Looney E. Wood** is new chief engineer of Bristol Aviation's Tyler Instrument Division. **J. E. Clark** and **Kenneth C. Baylock** have been made chief engineers of Chicago Vought Aircraft division.

C. A. Hoffmann has been appointed manager of a new industrial blade sales dept. of **Continental Tool & Machine Co.** He will function in part of the force available products division. **Chas. T. Langston** has joined the mechanical engineering dept. of **Hughes Aircraft** as design staff engineer, instrument.

With the **Julien-Capt. Joseph J. Keller** has been promoted to senior design operations manager for **Boeing Air Engineering Corp.** **E. H. Porter**, **William L. Monette**, **Frederick** have been named Edo's traffic and sales manager for New England.

M. B. Jones has been appointed Texas World Airlines' mgr of engine repair division. **Bernard R. Sorenson** has been named Canadian mgr for **Royal Dutch Airlines (RDA)**, according to **William DeWitt** who has been transferred to RDA's office in Kansas. **Robert F. Hahley** has been appointed general sales manager for **Citibron**, General Aviation and will handle sales development in CNA's divisions at Los Angeles, San Francisco, Oakland and San Diego.

Connelley-C. E. Stewart has been named as mgr of Northwest Airlines. He was previously identified in this column May 29 as a Western Air Lines official.

INDUSTRY OBSERVER

Air Materiel Command helicopter researchers are continuing their project for experimental towing of helicopters behind four-wheel planes to extend engine range, despite Washington reports (AVIATION WEEK May 25) that USAF considers the wrong speed too slow to be practical for low-rate recovery work. First report from tests flew has been confined to the Sikorsky H-33 in auto rotation. But project calls for similar tests eventually on the larger Sikorsky H-19 and the Pavesi H-12.

National Advisory Committee for Aeronautics now has a variable stable air research plane at Edwards AFB, Mo., Calif. USAF and Navy have turned over to the research agency the transonic and supersonic research planes originally produced and flown, including Douglas D-558-1 Skyward, D-558-2 Skyrocket, one of the Bell X-1 and the Northrop X-4. Plans are to transfer the three low-speed Bell research planes, X-2, X-3, and X-5 (Aviation Week May 28, 1949) will go through USAP flight trials at Muroc before NACA gets to fly them.

Rolls-Royce and Pratt & Whitney have come to a parting of the ways on Anglo-American turbine engine development. No further collaboration is expected after any finalizing details of the T-56 joint program are cleared up. The decision represents a change in plan, for Pratt & Whitney had previously indicated intention of continuing engineering cooperation with the British engine firm (AVIATION WEEK March 6) on subsequent proposals.

Carl Gustafson's Administration technical service is prepared to accept British certificate of transport jet engines, such as the de Havilland Ghost, if any American manufacturer wants to use them in its planes, but would still criticize the particular plane installation.

Curtis propeller division on Curtiss-Wright will soon make delivery on the axial engine for the Republic XF-91. The Curtiss engine for the XF-91 has four 4000-hp. thrust cylinders, while the Curtiss engine for the Bell X-3 develops a total of 15,000 lb. thrust from two cylinders, one giving 5000 lb. thrust, the other 10,000 lb.

Bell Aircraft Corp. is about ready to sell out the X-3 research plane, but is still waiting for the Curtiss engine. Bell's first X-1A, an unpowered version of the X-1, will be ready about next January. This is to be powered by a reaction rocket engine developing 5000 lb. thrust, as does the engine on the X-3, but using a turbo pump which will reduce weight of the engine installation and give a longer firing period.

The de Havilland Dove, twin-engine, eight-place transport, has received its U.S. airworthiness certificate and the Canadian de Havilland company is now expected to launch a strong U. S. sales drive. It will be aimed at the executive rather than feeder market. The Dove will sell for \$60,000, compared to something over \$70,000 for the Beech Model 18, closest comparable U. S. transport.

Establishments Peaga et de, France, has changed the name of its single-seat tailfin powered lightplane from Cyclone to Sylphe. But its new change is a prelude to a World Aeronautical Corp., sharing proprietary rights to the name Cyclone. Sylphe, undergoing low flight tests this month, is a powered version of C.M.S.11 tailplane aircraft former designated C.M.S. R-13, the plane was developed in a test bed for the turbojet. Peaga's turbojet engine, developing 170 lb. static thrust, Peaga weighs 120 lb. fully equipped. It is mounted in the Sylphe above the wing; part of it is the engine. Peaga has also under construction a two place powered version of its C.M.S. R-13 glider. It will be powered by a new Turbomaster radial flow turbojet developing 600 lb. thrust. Both single and two place planes feature tailfin, cantilever construction and butterfly tail surfaces.

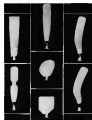
Societe Nationale de Construction Aeronautique du Sud-Ouest has entered the transonic jet transport development program. Engineering studies are under way using a modified version of the SO 540 derivative engine. The engine jet-engine transport accounts two Rolls-Royce New Mk. 5 outboard-flow turbines in place of engine Pratt & Whitney R-2500-B41 1600-hp. engines.



TEST BLADES of Hans Standard seemed may shapes. Left, shaft aerodynamic Group; Rome with mockback blade, one with some indication supposed to work.



off aerodynamicity and a swept slab one. Center shows the answer: this blade, upper bay. Right, more clear upper bay, this blade, top center, World War II



Male, top right, modern sweep back, one, sweptback blade, lower left, sweep indicator; lower center, slab, lower right, center sweep back.

Props to Drive Planes Faster Than Sound

Tunnel tests offer conclusive proof that supersonic propeller flight not far off.

By Alexander McSweeney

Supersonic flight in propeller-driven airplanes—long believed physically impossible—now seems on the basis of wind tunnel tests. In the next two distinct steps, it will be a reality, bringing with it promise of important new economies in high speed flight.

What now being done indicates that about 15 months hence today the first supersonic turboprop-driven airplane may slice its way through what was once considered an insuperable sonic "barrier" to propeller aircraft.

Leading U. S. propeller designers today are confident that they have their all-ways engine, compressibility, on the major testing helpfully for the knock-out punch.

Today they are testing new ultra-thin straight blades in static model propellers with diameters of three and four feet. Hamilton Standard division recently disclosed that in high speed propeller model tests in United Aircraft's wind tunnel, supersonic speeds equivalent to Mach 0.92 (equivalent to 700 mph. at sea level) had been attained.

► **Mincks Well Helped**—It is understood that AeroProducts has attained very high speeds in tests of model propellers

Still to be accomplished before a supersonic propeller plane can take its place beside the Bell X-1 first supersonic plane (Aviation Week Dec. 22, 1947) and other later supersonic airplanes, are the following steps: 1. Getting some finished engineering and production models.

► **Full-scale blades**, at least as thin as the model blades, yet very wide of chord, and stout enough to take the air loads imposed at supersonic forward speeds, must be produced.

► **Problems of propeller gearing and controls**, under load conditions of the first turboprop engine, must be overcome.

► **The turbine engine** thrust must be developed to a point of greater reliability and to powers well beyond those of the first turboprop now being.

Development and research credit for the forthcoming supersonic propeller can be divided between the Navy, National Advisory Committee for Aeronautics and the three principal U. S. propeller manufacturers, AeroProducts, Hamilton Standard and Curtiss-Wright. The Air Force made the mistake of writing off the propeller as obsolete when turboprop power came along.

But, belatedly did the USAF climb back on the propeller bandwagon, when

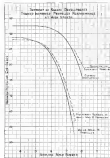
their error became obvious.

On a basis of turbine propeller now flying, AeroProducts has a lead over its competitors. Four inclusions from the General Motors division at Dayton are showing at the mouth of the Convair XP-57 turboprop engine test, while a fifth, six-bladed model the Douglas A1D Skyhawk attack plane up into the air two weeks ago for the first time. All these are fitted in T-40 double turbine propellers, built by another GM division, Allison. Yet another Aero-prop enters at the shaft of the big T-40s. A1D Skyhawk attack plane, developed by Northrop, but never left them.

Hamilton Standard has recently announced a new eight-bladed dual rotation propeller developed for the Navy, for turboprop use, which will undoubtedly figure in the high speed propeller future game.

Historically, AeroProducts had a four-bladed on the first flight of what is claimed to be the first turbine propeller plane in the U. S. S. to fly, the Convair XP-57 experimental Air Force fighter. A Hamilton Standard propeller was also and later on the plane. Another Hamilton Standard propeller was fitted to the General Electric TG-100 power plant in the nose of the Ross NF-104 experimental Navy fighter.

None of the propellers now flying are in the supersonic class, at least with their present power and their present plans.



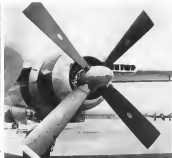
FROM THE of supersonic propeller is measured graphically as chart at left efficiency will level off at a high figure as the turbine runs, in contrast to the bell-shaped

► **Will It Be A1D?**—But perhaps, the Douglas A1D isn't too far away from becoming the first supersonic propeller-driven plane, as Douglas engineers recently hinted (Aviation Week June 19). It is equipped with a speed of around 550 mph. in level flight with its present engine/propeller combination. Engineering indications are that the plane with additional thrust, might be stepped up to speeds between 600 and 700 mph—under certain conditions right around the Mach 1 line, with additional thrust. Presumably this would be applied just by further improvement in propeller blade design, and partly by additional shaft hp. from improved turbines.

To appreciate fully the radical change which has taken place in propeller design theory, you must remember that until recently, the best propeller engineers put the top limit for a propeller-driven plane somewhere around 300 mph.

Actually about the top speed propeller-driven fighters of World War II were able to attain in level flight, was right at the 300-mph mark. The same models was flat with the propeller blades of these drive efficiency began to drop off rapidly as the plane speeds approached beyond the 150-mph mark.

This was because the propeller tip was moving at speeds faster than the airplane's forward motion. (Rake tip speed is made up of two factors,



ated for both forward and cranking configurations. Propeller at right, Hans Standard's Tachometer, at one stop measured from the engine prop, being comparable

forward speed of the airplane, plus the rotational speed of the propeller.) And at such as the top speed got up into the transonic regime, the blade ran into compressibility troubles, which became worse and worse as the speed increased.

If still more power was added, the shock waves would burst up the propeller into a vortex, the engines can die.

► **Why Be Better?**—But why is it then, that such a big jump in propeller air plane speeds is now confidently forecast by authorities, such as Dr. Hugh L. Dryden, research director of NACA, in Congressional testimony, and Rear Admiral C. M. Bohrer, Navy Bureau of Aeronautics assistant chief for research before the Institute of the Aeronautical Sciences?

The explanation lies in the aerodynamic research which has been applied to propeller blade designs in the intervening years since World War II's end.

The basic secret of making an airfoil which can slip through the viscous drag without paying out in expenditure of air at its leading edge, are now available to every top aerodynamicist. And they apply equally to propeller blades and airplane wings.

In aircraft terms, a supersonic wing or propeller blade can be shaped either by making it very thin, or by sweeping back its leading edge so that all the leading edge does not take the shock wave at the same time. And if the

to the "current development" in chart. This prop design has been test flown with conventional engines and is designed for both single rotation and dual rotation wing, or blade, in both thin and sweptback, as much the better as for its transonic penetration characteristics are concerned.

► **Straight and Swoop**—Swoopback was tried by most of the manufacturers, but has been largely discarded. Structural problems involved in making a swoopback propeller that can take the necessary loads are far greater than those of making a straight blade propeller.

And the same result can be achieved by thinning the blade. A Hamilton Standard swoopback propeller showed only moderate improvement in efficiency over a World War II propeller, and was much less in efficiency than a straight blade than propeller. (See accompanying chart.)

It is important to differentiate between a supersonic propeller and a supersonic propeller-driven airplane. Propellers moving at supersonic speeds are not new. Ever since World War II and possibly even a little before, propeller designers have been dipping into the supersonic speeds with their propeller tips. The big 18 ft. Curtiss propellers on the Convair B-36 have supersonic tip speeds. Some helicopter rotor blades, too, have touched in the supersonic area, but getting the whole propeller to perform efficiently while passing a plane through supersonic speeds takes more development.

How thin will the blades be? Current blades for high-speed propellers now fly

are only about half as thick as the blades of the World War II propellers, and they possess relatively high propeller efficiency up to around 600 mph.

To achieve maximum efficiency at around Mach 1, George W. Becht, Cornell University physicist, estimates that the thickness-to-chord ratio of the blade of the tip must be about 7 percent as compared to 8 to 10 percent ratio of today's conventional propellers.

George Rana Hanelius Standard Aircraft Manufacturing, predicts increasing effectiveness of propeller design, stating that will accompany the development from here on.

There will be lifts, if any, change in the weight of a propeller of given size, with the change in blade thickness being the principal structural change. There will be increased effort to design more closely to the ultimate capacity of a propeller blade, with greater attention to analytical methods, and study of where the loads will eventually be on the blade.

► **Simplification Seen**—Once propellers are developed to get through the transonic range, considerably there may be some simplification of propeller design.

Heretofore with a leading cambered propeller, the design of the propeller, if the propeller is to absorb increased power by adding more propeller blade area. This led eventually to long-blade and later to an blade and right-blade dual rotation propellers. However, if the propeller can be turned more rapidly without great efficiency loss, a smaller blade area will absorb equivalent power so that there will not be need for dual rotation propellers in some of the power ranges where they are now required.

There is also the possibility of cutting down the rotation rate into now needed for propellers. For turbines, currently, the ratio is something like 10 to 1. But if the propeller of advanced design can be rotated more rapidly than now, the ratio might be cut down eventually to as much as 5 to 1 with resultant saving in weight and cost, equipment profit.

Mass economy of the propeller drives airplane however at its efficiency at low speeds for takeoff and climb, as compared to the turbojet airplane. And it is here where the economy of the high-speed propeller, which is expected to make itself felt most.

At any speed below 575 mph, the propeller-driven plane is more efficient than the jet-powered plane. And the slower the speed, the greater the advantage on the side of the propeller, thus increasing the economy of the propeller plane is equivalent to the thrust-on of the jet plane.

Anti-Sub Copter

Bell wins competition for Navy contract with tandem rotor design.

Bell Aircraft Corp. last week came out winner in a hotly contested race to build a new anti-submarine helicopter, to insure the Navy's contract for development of a big new anti-submarine helicopter.

Contract for production of three experimental prototype helicopters plus the engineering was unofficially reported as close to \$5 million. Negotiations for the contract have been authorized, now that the competition has been decided. The new machine will be twice as big as anything Bell has yet built, weighing 11,000 lb gross weight. It will be powered with a Pratt & Whitney JT-300 engine rated at 2300 hp. It will represent the first departure of the Bell organization from the light helicopter and sub-rotor configuration, to the two-tandem rotor configuration, which appears to be catching on as the new type is larger rotor wing aircraft.

Bell's plan to build a tandem rotor copter for this competition was first disclosed in AVIATION WEEK April 10.

The anti-submarine copter is expected to be much faster than the current crop of helicopters and capable of carrying sufficient anti-sub weapons to be a deadly killer of the modern craft.

Bell has built approximately 300

small helicopters for commercial use, and a considerable number of two-place military craft, as well as 11 of its larger YH-130A machines for the Air Force. The company also has plans for developing the 600-hp YH-130A, which has a 6150 lb gross weight, into a commercial passenger craft designed to carry 11 passengers with an 800-hp engine.

To win, Bell had to beat out victoriously eight recognized helicopter firms in this country. Competitors included Pavesi, Sikorsky, Cessna-Wright with a Dornier design, McDonnell, Kellner, United Helicopters Inc. (UH-1), Kaman, Gyroplane and Hughes Aircraft. At stake was one of the greatest contracts yet available for the struggling light helicopter industry. It will give Bell a strong advantage in development of both military and commercial craft. Some other designs entered, included such radical devices as Pavesi propellers at the tail to increase forward speed, and sub-rotor loading, another device in getting higher speeds.

Johnson Plane Ban May Kill Air Races

U.S. Defense Secretary Louis Johnson has an unlikely change of heart, the 1950 National Air Races at Cleveland will probably be killed off.

Frank Crawford, National Air Races President, told AVIATION WEEK in Cleveland last week that Johnson might change his order disapproving participation of Air Force and Navy planes.



A THREE-ENGINE PICTURE

copter replace its earlier model Bell which had been in service two years. The New York Police Department will use them for street service in traffic control, search and rescue.

At the 1950 races, Crawford and Johnson had told the Air Force officials, "If the Chiefs of staff ask for permission to send the planes to the races, then that will be a different matter."

► In Washington, A spokesman for the Defense Department had AVIATION WEEK that the matter was not scheduled to come before the Joint Chiefs of Staff and that the subject is closed. Johnson's letter to Crawford on the subject was explicit: "For reasons of training and economy I find it mandatory that no aerial or other forms of public military demonstrations" be held.

For the last several years the dual structure of the races, the local high speed flight and precision aviation of the latter combat aircraft as well as the state display of virtually all of the nation's modern night industry-sponsored races, which rounded out the close, have been particularly divided in the split in speech between military and civilian aircraft officials.

► **Disagrees Rule**—Military aircraft have experienced some times in military events with steering planes in the close spectacle of the air show. Military pilots in making speed runs across the air race grounds have, on several occasions, nearly wiped out the civilian and military wings apart the delicate control of their planes appearing only a few feet above ground.

For the last three years the Navy and Air Force have annually considered withdrawal from the races. Value of military participation from a publicity point of view was weighed in light of the overall costs of participation and the hazards of the spectacle.

► **Cost—Race officials**, last year, paid over \$40,000 to deliver costs of USAF, Navy, and National Guard aircraft operating in the show. That amount, however, was only a fraction of the actual cost of participation. Race enthusiasts and some military officials have attempted to lay the burden of the costs of military participation to the races.

Aircraft industry executives concerned with keeping their planes favorably in the public eye have suffered severely over the prospect that one of their planes may crash out of public favor. For example, two years ago, after the major explosion of a Lockheed F-80 jet beginning its steep climb the airport was during the All American Air Meet at Miami, the Air Force was delayed with serious personnel and equipment in ground and the Lockheed F-80, in particular.

The Canadian government, which each year has sent a delegation of pilots and planes to the races, will also probably decline participation in the air show. A substantial number of the planes is being loaned out on the basis of a lease that provided over 1949 F-80 Shooting Stars.

the races personally only as a "hands-on-the-leader" gesture.

Only about currently out so far, for the year's show, are the major series (100 in all, approximately) which will be sponsored by Government, Military Corp. Five major for the race duration and a special consolation prize series at \$15,000. This also was started in 1947 and was sponsored for three years by Goodyear Tire & Rubber Co.

Liaison Competitors

Hit Cessna Choice

Piper Aircraft and Taylorcraft, Inc., have filed protests on Cessna's decision as to produce through USAF up to 500 Cessna L-19 (Model 385) liaison service aircraft. About one-fourth of the planes will be assigned to National Guard use.

The Army decision was based on the recently completed aircraft competition between two helicopter manufacturers at Wright Field, Ohio, and Ft. Russ. N. C. (AVIATION WEEK June 5). Piper and Taylorcraft, letters to Secretary of Defense Louis Johnson, both claimed that Cessna Model 385 exceeded by nearly 900 lb. Army's competition requirement that weight not exceed 1800 lb. Piper PA-39 engine, weight is 1870 lb. Taylorcraft Model 18 engine weight is 1080 lb. Cessna's engine featured a 211-hp Continental E-160 engine while the Piper



F-34 LINEUP

New Lockheed airplane plans had resulted at the factory prior to delivery to the Air Force. The all-weather fighter has been assigned to guard the strategic airbase was of the U. S. Powered by an Allison turbine with streamlines, the F-34 was a pilot in the world's first and was in the air last night. A substantial number of the planes is being loaned out on the basis of a lease that provided over 1949 F-80 Shooting Stars.

and Taylorcraft aircraft were equipped with 325-hp, Lycoming engines.

Johnson has "ordered the letters, through channels, for investigation," according to an official spokesman. "Reduced number of all five entries—Cessna 385, Luscombe T-34-A, Taylorcraft 85, Piper PA-39 and Pieten 11-121—will be conducted by USAF at Wright Field. Technical evaluation of all competing planes was conducted by the Army of Ft. Briggs.

Cessna's L-19 is a high-wing, all-metal plane with 35-hp wingspan, empty weight of 1440 lb., maximum speed of 130 knots, cruising speed of 50 percent of maximum, and maximum speed of 43 knots, and service ceiling of 22,000 ft.

House Passes NACA Wind Tunnel Program

House of Representatives has given National Advisory Committee for Aeronautics a good push toward its \$46,000,000 supercritical wind tunnel program. The program would provide better facilities for research and results "in the speed ranges from two to five times the speed of sound."

The House approved a \$75 million cash appropriation to launch the program immediately in a 1950 fiscal year deficiency appropriation bill, still subject to Senate action. The President had requested only \$5 million cash and \$15 million in credit authorization.

The House was quick to point to the urgent national defense need for high-speed wind tunnels and the fact that stopping up the program, now spending over a million per year, would result in a reduction of overhead and other expenses.

With \$75 million cash, the House said, it will be possible to complete three high-speed wind tunnels, under the seven-year program, would cost \$102,244,000, as follows: An eight-foot tunnel at the Ames Laboratory, with an originally estimated cost of \$12 million; a four-foot tunnel at the Langley Laboratory, originally estimated at \$10,917,000, and an eight-foot tunnel at the Lewis Laboratory, originally estimated at \$44,341,000. The eight-foot tunnel will go up to approximately 3.3 times the speed of sound, and the four-foot tunnel will reach a speed about five times that of sound.

The President's \$20 million estimate, spread by the House's Representative, had originally estimated \$15,500,000 for supercritical wind tunnel construction on the eight-foot Ames tunnel, and \$30,000,000 for design work and land purchase for the four-foot Langley tunnel.



NORTH STAR with fan growing out of its back is used for RCAF icing research. Scientists observe fan in flight, testing icing formation.

How RCAF Weather Researchers 'Break the Ice'

St. Hubert's RCAF Base, Quebec—What is better? Electric current or hot air?

RCAF scientists have proved to their own satisfaction that for wing de-icing, and most other de-icing problems, the electric current system they have developed is a better "deal." At second, then the "belonging" de-icing system now used on most modern transport and military planes.

► **Flying Ice Wagon**—Donald Fraser, leader of the RCAF cloud physics project at Rutherford, explained to Aviation Week the new "flying ice wagon" being used in the project. The plane was on display at St. Hubert's RCAF Base in Quebec, Canada.

The four-engine Bell-Boeing Merlin-powered Canadair DC4M which is typically equipped for wing icing research, doesn't look like any other DC4M you ever see.

It has a large square vertical fin, an extra cone, about midway back on the fuselage, which gives it a shark effect. It has transparent plastic bubbles for observers on both sides, behind the control cabin. The wings have strips of electric wiring running along the leading edges, and additional electric wiring further back in the top wing.

The big fin, used for wing tests, is also used for de-icing, as are the propellers.

Fraser said that findings of his research group had been subject of disagreement by the U.S. wing research group at the National Advisory Committee for Aeronautics. But only recently, he added, did the NACA directors find evidence that they were coming around to favor the electric system as well.

► **New Technique**—Technique used in electric de-icing calls for instantaneous applications of current to disintegrate ice at intervals, rather than a steady flow of heat as is supplied by the hot-wire system. While electric de-icing in popular newspapers, the other applications are experimental.

Fraser said the test rig used near the leading edge, above and below the narrow wing strip and at the brief application of current to the wing further back on the wing, broke up the frozen rain and cleared the wing. Power is supplied by means of two generators with airfield engines in power supply. The power amounts to 60 kw.

► **Observation**—The propeller observer has a test behind the fin on the fin where he can see the propeller wing

conditions, through a stereoscopic viewer, or by stopping the propeller. A stereoscopic observer, usually Fraser, charts the propeller's flight in a search for the proper details to get wing conditions. A search radar is available, not yet completed, will later be used to assist him in his search for details. He also studies ice formation on the vertical fin windshield as the fuselage.

The wing observer similarly studies wing wing formations, while an observer observer keeps an overall check on the thermopiles and other equipment used for scientific measurements of the test.

Fraser said that in the tests so far, the plane has experienced temperatures as low as minus 9 Centigrade, and has flown at altitudes up to 15,000 ft. The flight research data, he said, were a useful to earlier ground test chamber experiments which had pointed to the electric wing system as the most reliable means of "breaking the ice."

The RCAF's Ice Wagon will continue its icing research flights until the project group is satisfied that it has completed a complete set of data on wing phenomena, under various atmospheric conditions.

► **Noticed Airlines**—Purchase of 600 common shares by Joseph Moroz, Jr., making a total holding of 5100, purchase of 1170 common total holding, by William A. McCarry, director, purchase of 3000 shares, holding, by Paul R. Scott, director.

► **United Air Lines**—Sale of 170 preferred shares by John D. Davis, director, making a total holding of 1040 common shares.

New officers reported company holdings as follows:

► **Flightline**—Director and Airplane Co. Fred Bennett, Jr., no holdings.

► **Northwest Airlines**—Kenneth Brown, no holdings.

► **Northwest Airlines**—G. L. Stewart, no holdings.

PRODUCTION

Minimum Wage Fixed at \$1.05

Labor Department decision splits difference between 95-cent industry figure and union demand of \$1.15.

A minimum wage of \$1.05 an hour will be required from the start of the busy for work on most government contracts signed on or after July 8. (The new law is forecast in Aviation Week No. 15.)

Secretary of Labor Maurice F. Tobin, in his introduction of the Wage-Hour and Public Contracts Administration, has determined \$1.05 to be the "pre-swing minimum wage" as a result for the purpose of the Walsh-Healey Public Contracts Act. This law governs working conditions and minimum wages which must be paid for work on government contracts amounting to more than \$10,000.

► **Maximum Doubled**—The same order also doubles the 50-cent maximum which has prevailed heretofore, though not retroactively, in the aircraft industry since 1938.

The definition of what products are included in the new regulation is also revised by the order. Last revision of product definitions occurred in 1942.

The new definitions consider procedures begun last year when the Civil Aeronautics Board petitioned for raising the maximum. At hearing held last July, the union and two others argued that the higher rate should be \$1.15, based on a survey made by the Bureau of Labor Statistics of 14 aircraft plants employing 165,000 workers, exclusive of bonuses and overtime.

The Aircraft Industries Assn. opposed any wage devaluation at the time because of inability of wage and non-wage prices and the "anti-federal established plan" of aircraft wage rates. It also opposed reducing the industry overall. If new wages are made but to be made, AIA suggested a minimum rate between 50 and 95 cents an hour.

► **Split the Difference**—The Tobin \$1.05 rate seems to have split the difference between the top figure cited by the industry—95 cents—and the \$1.15 cited by the unions. (Tobin originally cited upon a minimum wage of \$1.02, but the CIO submitted new arguments which finally persuaded the secretary to add three cents to the figure he had already decided upon.)

The figure from the figure from statistics is 81% or

Since 1915, at 82 percent of the 140

aircraft plants covered by BLS employ 14,800 workers earning less than the \$1.15 rate the unions wanted. These 115 plants employ 115,400 of the 165,000 employees covered in the survey. This indicates, Tobin's order points out, a "working minimum wage" below \$1.05.

On the other hand, more than twice as many employees covered by the survey earn between \$1.05 and \$1.09 an hour than that. Half of the plants surveyed, employing more than 50 percent of their employees earning under \$1.05.

Some plants have no employees earning more than the higher minimum. Tobin's order says it is impossible to estimate how many workers will get wage increases under the new order in how much the maximum will cost in higher wages and higher cost to the government. But all workers in a plant fall under the higher maximum just their already working on the government contract.

A lower minimum rate for apprentices was set at 75 cents. No provision was made for learners' rates. They mean that learners, too, must be paid at least \$1.05. Subminimum rates for hands engaged workers will be set, an employer request under regular wage law practice.

The aircraft industry industry made clear that all electrical equipment not just a few specified items, as excluded from the industry.

Production of engines for gas turbine and electrical accessories also continues included in the industry, but production of electrical components themselves was excluded. Electrical items which produce electrical electrical aircraft gas turbines and accessories included on this work be excluded because it was essentially electrical.

Tobin refused to exclude aircraft parts from the industry because they are made primarily for aircraft maintenance. But he excluded pumps and valves primarily made outside the industry which manufacturers install in aircraft parts.

The \$1.05 rate is the same in that the workers and aircraft industry. Only in steel has a higher rate been set—\$1.21, \$1.09 and \$1.08, depending on the region.

C-W and Doman Clinch Copter Deal

Cantus-Wright's plan to enter the helicopter field (Aviation Week No. 14) slipped up once stronger last week when Doman Helicopters, Inc., notified its stockholders that it had concluded a five-year engineering service and royalty agreement with C-W.

The contacts gave Cantus-Wright the right to build and sell helicopters of more than 750 hp incorporating the Doman-engineered rotor system. Furthermore, C-W secures the engineering services at Doman and all patent and future patent developments for the term of the agreement.

Doman is completing its eleven-place prototype Arctic rescue type rotorcraft L2-4, which is expected to be test flown in about 30 days. Following initial tests, the L2-4 design is intended to be developed into a production model.

► **Doman to Up Stock**—At a special stockholders meeting, scheduled for June 5, stockholders also will be asked to vote on an option of Doman to purchase 200,000 shares of common stock at a price increasing from \$2 to \$6 per share over a five-year period starting April 27, 1950. By July 1, the company's treasury and payments of stock to secure various contracts, Doman's personal holdings had declined by about 70,000 shares by the fall of 1947. Replacement of Doman's stock earnings would guard against control passing to another group and serve as a consideration for his acquiring the company had been by Cantus-Wright in the new financing agreement.

PRODUCTION BRIEFING

► **Glass L. Martin Co.** plans among in cost and engineering time by using aluminum in the construction of its new aircraft to produce light weight structure for its current production method of casting tools or templates.

► **General Electric** plans to spend \$1 million on altering the Tashkent, Miss., plant of its plastic division. Manufacture of plastic shells to be used at Tashkent and Dugway, Ill., will be concentrated at its Pittsfield, Mass., plant. Several chemical manufacturing activities will be expanded at the latter location.

► **Memphis Howell** Regular Co. has moved its Memphis operations to a longer leased-facility from Northwest Airlines at Wolf-Chamberlain Inc. Field. The structure measures 200x150x50 ft.

SEC Reports Stock Transactions

Sole by Airfrat, Inc. of its total holding of 100,000 common shares in Piper Aircraft Corp. is reported in the latest Security and Exchange Commission transaction report.

The sale of 2000 common shares by William Thomas Piper, officer and director, leaving a total holding of 146,270 shares and 9950 preferred shares was also reported.

Other aviation transactions listed for the period from mid-April to mid-May were:

► **American Airlines**—Sale of 200 common shares by Charles Glendon, director, leaving a total holding of 480 shares.

► **Boeing Airplane Co.**—Purchase of 200 common shares by William Allen officer and director, making a total holding of 1751, purchase of 286 common shares by Fred F. Laidlaw, officer and director, making a total holding of 361 shares.

► **Capital Airlines Inc.**—Purchase of 400 common shares by George Shaw, director, making a total holding of 1300 shares, purchase of 1200 common shares by Raymond Leitch, director, making a total holding of 1410 shares, purchase of 1000 common shares by Charles Henderson, making a total holding of 6000 shares, purchase of 1000 common shares by Robert Wilson, making a total holding of 1076.

SALES & SERVICE



Toll of Weather and Mountains

CAA report on non-air carrier accidents shows that pilots are still disrespectful of instrument conditions.

"Continued wind flying into instrument weather," became the homocidal obituary for 233 U.S. private pilots in 1948 and 1949.

Although fatal accidents in 1949 were 550, a 35 percent decline from 1948's 836, accidents attributed to "weather" were more numerous than in 1948, representing over one-fourth of the total fatal accidents in 1949. "Pilot error," including weather accidents, misreadings and failures to maintain flying speed, accounted for 90 percent of the fatal accidents which occurred in 1949.

This season is noted in a new report, "Fatal Accidents and Weather (Non-Air Carrier) Calendar Years 1948 and 1949," compiled by Civil Aeronautics Administration's program planning staff of the aviation statistics division.

The national average figure doesn't tell the whole story. The difference in rates between states was wide. During 1948 in Connecticut and New Hampshire, weather is blamed for 50 percent of all fatal accidents, the total rose to 40 percent in New Jersey, 33 percent in Maryland, North Dakota and Utah, 31 percent in Pennsylvania, 20 percent in California, New Mexico and North Carolina, and 20 percent in Massachusetts.

Mountainous areas, as might be expected, figure prominently in the accident picture. The accompanying chart, if superimposed on a topographical map,

would clearly show how hazardous these areas are when visibility is low.

■**NC Fades.**—Breaking the score down still further, in 38.5 percent of the cases the fatalities occurred in states other than the one in which the aircraft was registered. In fact, including accidents involving pilots operating in similar portions of their own states, perhaps half the fatal "bad weather" accidents can be pinned on planes operating in strange territory. In 19 of the 49 states, including the District of Columbia, at least half of the weather fatalities involved planes with out-of-state registration.

Even some boasting combat weather merit of the team, such as California, Texas, Oklahoma, Utah and Nebraska, may also rank high in number of accidents. This seems to be due in some measure to owners in these states discounting the weather factor. They license but in checking reports to obtain pilot licenses for subsequent, though potentially dangerous, unfavorable conditions.

That too few pilots checked weather forecasts is implied in the report that flight plans were filed in only some of the 113 fatal "weather" accidents that happened in 1948.

■**Pilot Ratings.**—Those holding private pilot ratings achieved the highest honor of better average than the largest number of accidents—478, or 70 percent of the total. Commercial pilots were less

valued in 63 ratings (43 percent), and accidents totaled with 14 accidents.

The most striking statistic is regularly experience is the one showing that only 1.5 percent of the 233 pilots killed in accidents attributed to weather held instrument ratings, and none of even the total number who was qualified to fly IFR had that flight phase. None of the victims involved in 1948's 113 "weather" fatalities held instrument ratings.

BRIEFING FOR DEALERS AND DISTRIBUTORS

■**Easo Directory.**—A listing of airports in 26 states is now being distributed free by Easo. Given as guides of fuel location, class or one of the fields, and about they are open for service. Write for the Easo Co. Pilot, Easo Standard Oil Co., 50 Rockefeller Plaza, N.Y., N.Y., or any of Easo's various sales divisions.

■**Sales Pigeon.**—Southwest Automotive, Dallas, is using a little cartoon cartoon character "SAC Sam" to call customer attention to the firm's newest maintenance service. Employees hang the cut-out on customer plane dash. Small cards describing services are mounted in a slot behind "SAC Sam." Numerous inquiries are credited to the device by the company.

■**Miller 360 Pigeon.**—New performance figures for the 1950 Miller 360 engines give the engine a new rating, speed of 64 mph, weight 210 lbs., maximum mix of 4900 ft/min, and increase output of over 13,000 ft.

■**Hot Weather Tip.**—With warmer weather hanging on, increase in flying operation are mentioned to caution pilots that temperature increases affect plane takeoff rates and useful loads should be carefully calculated if the aircraft are to be handled safely.

■**Operation Miami.**—Operation and users of small airports throughout Tennessee are scheduled to get together in Knoxville on Aug. 19 for the Tennessee Air Progress Conference. The Southeastern Airport Managers Assn. is planning to hold its fall meeting in Montgomery, Ala., Oct. 10-11.

■**New Cessna Dealer.**—Glenwood Aircraft, president of Glenwood Business Aviation Service, Tulsa, Okla., has been accredited as a Cessna dealer for a large portion of eastern Oklahoma. Rossini is said to be using a new plane financing plan that cuts the cost of excessive insurance.



The Birdmen's Perch

Does your woman join a flying club?

That's a fair, high topic subject! But, whether you're considering the Associated Flying and Chandler Club, or the



Perkins Place of Progress Society, check these two points carefully.

(A) Is the club on incorporated club?

(B) Does it carry adequate insurance?

To be unincorporated club, should one of the "good and true" run a larger's line, you could be held personally responsible for damages needed by a court against the club.

If your club fails on carry adequate insurance, one experienced aviator could backstop the organization?

For your own protection, check the

financial assets of any flying circle before becoming a member of it?

AND—

As long as you're checking about joining clubs, join the modern parade of plane-happy pilots who wouldn't consider riding off without a canister full of Gulf grade Aviation Oil—Series D.

It's a super dope tale, the world's finest detergent detergent oil for horizontally opposed engines!

And, that's important. Gulfgrade Aviation Oil—Series D—is the only aviation oil put through Gulf's exclusive Alkylar process to remove that extra carbon/oil clogs factors.

No wonder tags and valves may fly free! No wonder pilots treasure the pounds between overhead up to 1000!



Call it insurance, if you like—engine insurance that isn't just a promise, it's trouble-free flying!

LITTLE KNOWN FACTS DEPT.

How many men have you met and to yourself, "Gee, I wish I was fitter!" Well, you can be. But, if you'll only read and take heed!

Today we offer our hints to a spectacular success of the "Miles Cap." M. W. 1200.

R. Coffin of 2005 West Seven Mile Rd., Detroit.

Bill a L.K.F. with 78000, once had the cherished Perch Plan (in Commemorative), with public acclaim, and possibly a great success in sales if he shows this to his boss.



The Lockheed T-96 Fighter not only has two engines, but duplicate sets of controls and navigation instruments, allowing it to sustain direct hits and still return to its base!

Whisper! Tell the pilot pool out! Your Commemorative is on the way. As for the rest of your shop.

Don't reveal strong features! Seal your Lips! Knows Plans About Will Known From—with FIDOT—to the address of Gulf Aviation Dept., Gulf Building, Pittsburgh 16, Pa.

Gulf Oil Corporation... Gulf Refining Company...



AVIATION PRODUCTS



AERONAUTICAL ENGINEERING



METLBONDING Film machine dips 1-ft. wide strip of Nylon cloth into M1 cement, copolymer solvent with heat, rolls in MGC cement and solvents. Finishes in 1 min.



CLEANING process for magnesium alloy requires wiping with low-flash naptha. Primer previously applied is not softened by low-boiling cleaners in used.

Metlbonding Saves Time, Money on B-36

New Convair adhesive development speeds production of 5500 sq. ft. of superlambar's skin assemblies.

Metlbond, developed by Consolidated Vultee Aircraft Corp., is believed to be the only adhesive used in the U. S. to join metal parts over large areas of a production airplane.

Convair's use of Metlbond has gradually expanded as the process has been improved, until at the present approximately 5500 sq. ft. more than one-fourth of the B-36 bomber's exterior surface consist of assemblies which have been attached to stiffening members and doublers by the Metlbond process.

Yet Metlbond has not been developed to its ultimate stage. Continuing laboratory investigations are expected to result in refinements that will make it more useful and economical.

Other Users—Metlbond has recently been made available for use at other manufacturers under Convair patents (inquiries have been received by Convair from Boeing, Douglas, North American, Chance Vought, and Bell in the aviation field, and from General Mills, Chrysler Foods, Princeton University, Arnold Optical and several other companies within the past few months). A number of these companies are currently analyzing the process and conducting tests to

determine the applicability of Metlbond for various uses. It is understood that Boeing and Douglas are further along than the other aircraft companies in their studies of the new process and its applications.

The process is comparatively simple. Surfaces to be joined are cleaned with a solvent. They are then covered with a thin film of cement. This is air dried for a short time, after which adhesive in film form is placed along the joining surfaces. Pressure and heat are then applied to the areas in contact to complete the bond.

To date, Metlbond has been used primarily on clad aluminum alloy and on magnesium alloy. It is applicable also to ferrous and to other nonferrous metals, but processing methods for these have not been fully investigated. Materials which need further investigation prior to extensive use of Metlbond include casted aluminum alloy, base steel, plated steel, thermosetting plastics, wood, and rubber.

Metlbond is used on magnesium alloy sheet, where design or production improvements result. The combination of materials creates as difficulty.

More Adaptable—The Metlbond process has a significant advantage over most of the other high-strength metal adhesives in that it is more adaptable to production use. Cleaning methods are not strictly critical, since ordinary solvent cleaners have been found satisfactory. Application in dry film form leads to simplification. It avoids many troublesome variables which occur when adhesives are applied by brushing or spraying. Curing temperatures and pressures are relatively simple to maintain within required limits.

One of the fundamental advantages of this process is that bonds can be made at pressures below atmospheric pressure on all sides, they do not have to be made to prevent the deformation which results from employing atmospheric pressure by applying a partial vacuum under rubber blankets. Since such tools are subjected to atmospheric pressure on all sides, they do not have to be massive to prevent the deformation that would be caused by the application of mechanical pressure, which for other metal adhesives must frequently be 100 psi or more.

Many applications for metal adhesives like Metlbond exist in machine construction, such as the build up of structural members to avoid tapered loading.

Process use of Metlbond is with this



SPRAYING of two thin coats of M1G cement follows cleaning. Laboratory tests now being made may eventually eliminate this phase of Metlbonding.



FILMING is shown being done on large blocks of hot section members. Big blocks absolutely prevent uniformly on parts when rubber blanket is used.



CURING wires is shown open to process loaded fixtures. Parts are bluelisted, cured at 570-590 F. for 25-37 min. at low pressure.



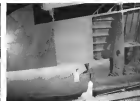
LOADING assembly in pressure fixture requires extra blocks for double edges. Loading jig, shown inserted here, must be flexible for various loading of sheet parts.



REMOVING the metal assembly from pressure fixture is quickly done, although care in handling may require steps is necessary to avoid deformation due to temperature.



INSPECTING of B-36 elevator panel includes visual check for voids, delaminations or poor bonding. Magnesium alloy parts get dry coat of zinc chloride after inspection.



B-36 RUDDER, here shown partially completed, shows interior and exterior appearance of Metlbonded parts. Smooth outer surface points up nonpenetrable glue from being process.

Bendix-Skinner

ORIGINATOR OF MICRONIC FILTRATION

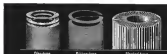
Results Prove It
the *Finest*

Name in Filtering

Now three out of ten Bendix-Skinner filters will apply the "finest" answer to your filtration problems. Here are the facts: available with patented, exclusive, twin segmented cellulose elements, simple, quick replacement, high flow rate with minimum pressure loss over 350 models providing filtration from 1/4 micron (.000019") upwards of flow rates from 1 to 5000 gpm. Why not let Bendix-Skinner filtration engineers work with you? Write us today.



Skinner-type is easily adaptable for operating at 1500 psi and 350 gpm. All models operate at 150 psi.



SKINNER FILTERS DIVISION OF
THE TRIMET AIRCRAFT, FILTER & EQUIPMENT
CORP. 2401 North Lincoln Road, C. Box 10, B.Y. 6, N.Y.
Bendix
SKINNER ASSOCIATES

pays through the required processing unless Method B is used rather extensively on the assembly.

Undoubtedly in the new art of joining structures with Method B and other metal adhesion becomes easier for designers, they will take advantage of its power, worth, and varied changes in some types of aircraft structures may be expected. Such structures may show a marked decrease in the number of component parts required. Lower cost production of airplanes with high-production assembly methods, and longer service life with less maintenance expense may be expected.

Tool Life Tripled By Special Process

A method of greatly increasing the life of high-speed cutting tools is presently being developed by Solvmetec Laboratories.

The process is called Solventizing. It is used only with tools made of high-speed steel that are already hardened, tempered and finished to size. According to the firm, tool life is increased by changing the characteristics of the steel to a "reasonable depth" to attain the qualities desired.

The company says that mechanical testing process does not bring about any dimensional changes, distortion or warpage of even the thinnest sections. There is no after-treating grinding. Under the patent arrangement, tools are sent directly to Solvmetec, treated there, and then returned to user. The operation is set up for 24-hr. service.

It charges \$1.75 for treating standard high-speed tools and \$15 per unit for tools weighing 4 lb. or less. There is a maximum charge on all work of \$75.

The firm guarantees a 50 percent increase in life of any high-speed cutting tool, but says actual experience—Adkins division of General Motors Corp. is one—report increases of 100-400 percent.

A Solvmetec executive told *Automotive Weekly*: "We can confidently guarantee our customers they will save thousands of dollars with our process, offering them less machine down time, increased production and a smaller inventory of high-speed tools. So far as we know, there is no other process like ours."

The firm is seeking interested companies to send a tool that is currently in use, or whose past production record is known. It will be treated and returned. If the user is not satisfied after 30 days, no charge will be made. Solvmetec says the process is the result of eight years of research and has been extremely proven. Address: Solvmetec Laboratories, 5918 Elston Avenue, Chicago 18, Ill.



Chance-Vaugh't's new NAVY F7U is powered by the WESTINGHOUSE J-34 Jet Engine. Turbine Control by HOLLEY.



HOLLEY
Turbine Control Co.

DETROIT 4, MICHIGAN

Data from British Ignition Forum

International experts discuss plug design, lead content, moisture effects, low vs. high tension, analyzers.

(McGraw-Hill World News)

London—The thorny problems of aircraft ignition systems were studied in a most two-day conference sponsored here by Lodge Plugs Ltd., Rugby. Attending were about 50 representatives of 15 airlines operating into Europe, and manufacturers of sparkplugs, ignition systems, and aviation fuel, and members

of government departments (Aviation Week, May 24).

The conference, first of its kind to be held in Europe, was modeled in a general way after the Glendon conference held in the U. S. last September. V. R. Banks, of The Associated Aircraft Company Ltd., presided.

The sessions were marked by a splendid frank give-and-take between the

operators and the manufacturers and covered many deep seated problems. The most of the conference was to be found in these exchanges. This is the first comprehensive report of the conference to be published.

Sparkplug Design

The conference seemed to agree that plugs should be designed for "low maintenance," i.e., at the least, should be good for the life of the engine-overhaul cycle without attention.

It was pointed out that ignition arcing required by the engine alone of all engine-serving parts. Most of the country's representatives by a good part of the blame on the plugs.

KLM's Nic. Lam indicated that he really would prefer to do an maintenance on plugs whenever, if he could get a 350- or 400-hour "reliable" plug at the right price.

• **Reliable Plug**—Lodge Plugs, according to Mr. Kenneth Hodges, its technical director, is working on the possibility of an "reliable" plug (its has a 400-hour life) with the manufacturer supplying out the scenery of platinum from scrap plug returned by the user for credit. (Lodge has recovered the metal from 16,000 plugs sent back after use by KLM.)

Mr. Hodges thought the whole idea of "reliable" plugs would be necessary if plugs were properly cleaned. This means that they must be designed in that they could be properly cleaned.

Trans-Canada Air Lines' R. W. Farnes suggested that consistent performance was more attractive to him. Thus engine maintenance as a whole could be better controlled and efficiently planned. He'd settle for a plug good for one complete engine-overhaul without maintenance. A 300-hour plug, he felt, was not good enough—he wanted a plug to "fit and forget."

It did not appear likely to him that a plug that would be fully dependable for the life of the engine period could be obtained while using the high-tension ignition system because of porcupine. High-tension present leads to the damage of plugs to increase their life and to prevent porcupine failures. These developments lay in three fields:

- Design of the insulator and top-ecity to discourage lead-leaking.
- Fine-wire vs. heavy metal electrodes.
- Making watertight the electrical connection.

• **Plug Tip**—He felt that the Lodge SR18/28 plug (standard on almost all European engines in whose principles it is applicable) represented just about the best of what the plug manufacturers could produce to meet lead-leaking.

In greater clearance between insulator and body provided the maximum (Continued on page 33)

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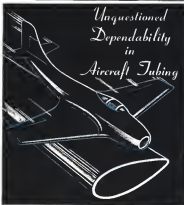
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opportunity for averaging action and also for the prevention and switching action of the wind, is windblast (also, its short wave action a lower acceleration gradient and hence less likelihood of cooling). From here, the prevention of lead loading was up to the engine manufacturer (to provide better circulation in the cylinder head, and thus better mixture distribution) and the fuel manufacturer (to reduce the lead content of the fuel).

Hopps and lead loading may be more common in certain in the Eastern Hemisphere than in U. S. service since U. S. aircraft fuel is lower in lead content (3.5 lb. of per imperial gallon, not exceeding 1.6 oz. whereas many British planes have to use fuel with as much as 4.8 oz. per imperial gallon). U. S. aircraft are generally operated on richer mixtures. This leaves pipework.

As to the electrode material, Hopps reported that since the U. S. aircraft had had no experience with the latest designs of British plugs, using platinum electrodes-American engine manufacturers had not improved them fully for use-the Americans still thought that nickel alloy points gave performance comparable to the best in the world.

P. A. Young of Wright Aeronautical, replied in Hopps' challenge of the American engine manufacturers, regarding the group that the engine builders had approved the Lodge plugs for service-test installations, but that so far no American operator has been reluctant in using extensive service life tests of these plugs. His company, he said, would welcome such test evidence-but would want to use the tests on at least a 500-plug installation here, and carried on to 500 to 600 hours of operation.

► **Free Wires-Martin Gunkem, of Pratt & Whitney,** said that with free wire electrodes there is no danger of per centage, and if fracture occurs the fragments fall away and there is no possibility of detonation.

TCA's Farnes reported that from service tests he knew of, free wire plugs in Pratt & Whitney 1830s in DC is fairly trouble-free as well, and plug troubles had been pretty high. By contrast, Whelan said electrode plugs were now giving 910 hr. service life when serviced regularly at 250-hr. intervals.

F. M. Soren of Lodge Plugs felt that what airline engineers want is the performance of platinum over while or through the heavy part of the solid electrode. He thought a combination of metals might do this.

Rolls-Royce's P. E. West suggested that he would like to see an entirely different ignitor design-as "invented" one-so that the lead deposited on the electrode will not run down to the tip of the positive electrode when engine

as in the conventional overtopping and instead shape design a studying such a design at the moment.)

As to non-ignition, Hopps felt that a design (such as developed in the U. S.) where the rubber hose belts against the electrode at the screen hole, should give satisfactory performance.

Cleaning and Maintenance

Gawing that some cleaning of plugs may be needed and the relative merits of sandblasting and liquid-cleaning were discussed.

Mr. Lam (despite his desire to do no plug cleaning whatever) reported that KLM's experience with cleaning showed that cleaning of plugs was only more tedious than new plugs. This, he felt, showed that his airline had developed pretty good methods for detecting a failing plug during overhaul-particularly in detecting cracked wires. But for the small plug, he admitted, "we don't really know much about what causes quaking failures." As much as a few plugs didn't give sufficient information, he said as operators must study a large number of plugs both in service and after removal-on only for the averages to mean anything.

Lam commented that the only proper method of cleaning a plug was to take it apart and clean the tip. But, on general grounds, he preferred to leave the plug assembled, sandblast it for 1 min. and not be overly concerned if a little lead was left in the cavity.

Farnes pointed out that sandblasting was 300 hr. where 100 hr. where deposits might be encountered, thus leading to tracking failures. He urged liquid cleaning, since in his experience a sandblasted plug was likely to fail in 200 to 300 hr. of service before of a new plug (as an average life) to 300% before (as an ideal life). He admitted that these differences might be due to any of several causes, and not put to the test that the plug had been sand blasted clean.

► **Cleaning China-Mr. Farnes described** the liquid cleaning method adopted by Northwest Airlines. This involves washing the plug for an hour in an 80% alcohol solution, then in a 10% solution of hydrochloric acid at 400 deg. F. for 30 minutes in a special washing drum. On its inner surface cones are fitted so that when the plug is only the upper portion of the plug is immersed in the bath, which is constantly agitated during the cleaning. Complete cleaning can be done or partial cleaning followed by sandblasting for a shorter time than would otherwise be called for.

G. H. Fletcher, of the Royal Aircraft Establishment, Farnborough, mentioned that the RAF has used ammonium acetate-with the addition of Tricel, a chemical setting agent-on a

non-ignited bath at atmospheric pressure, with some success, reducing the amount of subsequent sandblasting needed.

Hopps of Lodge Plugs had mentioned in the course of his paper on quaking plugs that the use of the sandblast and sandblasting in American plugs was less favorable for sandblasting than in Lodge plugs. He felt that American operators had really good sandblasting equipment, a fair chance. He disagreed with Farnes' remarks that sandblasting left pits in which lead would be deposited, the danger of the blast is so short that the course of the insulation is not damaged and the lead retaining characteristics are not increased thereby. If E. West of Rolls-Royce concurred with Mr. Hopps on this point.

► **Hot Ten-Farnes agreed** that some decrease in development for testing plugs, especially after they have been overhauled. Such a device would have to simulate the high temperatures encountered in service-ones a plug might use quite satisfactorily cold but fail after it had been installed in the engine.

Lam discussed the methods followed by KLM for testing.

► **Use of a psi stresser** devoted at the engine. This takes two hours to heat up a whole engine's array of plugs rapidly.

► **High-frequency radio waves** used for heating the plugs.

Both methods were so slow and so costly, he said, that he gave additional thought to his suggestion that he would preferably avoid a non-overhaul-bath for plugs, provided he could get plugs that would last the life of the engine.

Lead Content of Fuel

The problem of lead-fueling was probably the one approached as the results of most of those attending the conference.

J. G. Dawson, of Shell Petroleum's Technical Research, Glasgow, discussed how lead composition affected quaking plug performance. In the main he re-stated the instruction of the new version of last September's conference, showing how the decrease in lead itself and the use of lead in the fuel lowers the potential difference across the spark gap so that the quaking current is reduced to a point where it is insufficient to produce an arc across the gap. It is this short-circuiting effect which is the main source of trouble from lead fueling, he said, quaking being a minor trouble.

In dealing with lead-fueling, he said effective screening by the cylinder gases should be encouraged in two ways. This will not necessarily prevent the deposit of lead, but it will assist in disrupting



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the when it is deposited.

Dowson cited the graphs prepared by Deschamps at Pratt & Whitney as giving the most relevant data on one relating the amount of lead content in the fuel with the occurrence of lead poisoning. But the bulk of this data was obtained from engines on a laboratory test bed, and the two sets of data based on actual engine experience gave an entirely different slope to the curve. And the tests were carried out with a plug engine to be particularly susceptible to lead poisoning.

■ **More Pin-Leaf Leads:** The Associated Ethyl Corp's viewpoint was presented by F. K. Basko, the conference chairman. According to Mr. Basko, "you can have less lead in your engine gas-line if you don't avoid paying more for the lead." In support of a lower lead content, he said the Strattoners ran a little better on 108-115 fuel than on 115-145.

G. L. Shaw, of Shell, indicated that the trend in the amount of lead content in 108-115 fuel for European area is likely to be downward during the next 15 months toward 16 or 20 ppm impregnation. For the 115-145 fuel, however, the lead content is likely to remain constant for a while.

While a considerable reduction in lead content might be desirable, Dowson said it was generally regarded that only a small reduction would be possible. Similarly, a major reduction in the tail pipe and blow-by content would not be possible.

■ **Savagely Beat—**Unequal distribution of the lead in the cylinder—was one provision is made on the manifold design to prevent it—results in an unequal distribution of the lead for a scavenger. One of the most serious scavenger risks for fixed or turbocharged turbochargers, which, having the same solenoid as the lead, distributes itself throughout the engine in the same relative proportions as the lead.

Basko had this to say about the best low scavengers: "We've been looking for the right material, but we just haven't got it yet." He mentioned that most scavenger research had been carried out on a test engine, and the difficulties are that if the scavenger was any good in a scavenger, it was not chemically unstable that it wouldn't store it would "grab all the lead before the fuel was used."

The occurrence of production enter into it too he said. Only when it comes to a scavenger that is suitable for all from of fuel—as at least a large enough portion of them—it will be too accurate to produce in quantity.

Engine Analyzers

The new British Thomson-Houston analyzer, designed primarily for ground-

testing operation, system performance, was described by D. E. Welch. This unit, by means of ion-exchange, gives a simplified view of gas what goes on in each cylinder at the moment of ignition. By comparison with known "normal" traces, the engineers can diagnose what is wrong.

The British Thomson-Houston unit is not as suitable as the Speer analyzer in the breadth of information it can present, but it offers a useful assessment within its limitations.

Several members commented that while the high-tension "cylinder" might become quite familiar and the nature of the faults thus shown up quickly diagnosed, patterns from a low-tension engine would be much less distinctive and the diagnosis more difficult. More extensive training of the mechanics to recognize the faults in low-tension systems would be required. Rolls-Royce's J.E. West commented that the B-T-H equipment didn't distinguish sufficiently the nature of the fault and its location.

ICA's Purser reported that Pan American's fuel-bank with the airborne Speer analyzer had revolutionized the actual maintenance and inspection routine—no longer was the information as to trouble spots which the flight engineer could provide. This extended limited engine performance, to include propeller stresses, fuel system requirement, etc.

KLM's Lema said that his has been only just beginning to study ignition systems with the Speer analyzer on the test bed, but that already it had helped point out requests trouble or bad connections in the distributor.

Low vs. High Tension

KLM's Lema reported that his has had logged considerable time on the low-tension ignition system fitted to the Waspal RDM engines on the T-90 Cessna airplanes. Experience showed longer plug life and less lead contamination. A small amount of testing of the low-tension system in P-400 engines vs. DC-6s also showed increased plug life—up from 250 hours to 400 hours. As a result, KLM has decided to equip all its DC-6s with low-tension ignition systems.

KLM will next try out the General Electric high frequency (one volt) system on the Convair in about three months Lema indicated. He noted that his mechanics—being money-wise minded, as he put it—will probably be disposed to stick with the acceptable distributor design, rather than make the paper adjustments in the distributor. He hoped that the use of the engine analyzer (on the test bed) plus the switch to the high frequency system would reduce 80% of his engine troubles.

■ **Long Life—**ICA's Purser indicated



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FATEST THING IN FASTENINGS

the results of tests by Pan American which appeared to confirm that the Bendix/Knight low-tension system is P&W R-2500 engines had radically *confirmed* the claim of longer plug life. Life up to 100 hrs before inspection had been achieved, using a special Champion III-15 plug. The plugs were then reinstalled and run for as long as 1200 hrs.

Engine life shown in these tests averaged not more than 300 hrs. per 100 hr. Hence, with a gap setting of .015 and a heat of engine of .025, this would give a possible plug life of around 2000 hours.

Panes said the low-tension engine does not inherently increase lead-fading, but that the high-frequency system does. Welch of B-T-H commented that the high-frequency system increases the altitude at which the plug can be, with a larger gap.

Low Tension, Low Weight-Rates Ltd., according to one of its representatives, is looking for medium-frequency low-tension systems (2 hr., 250 volts).

Scintilla Ltd.'s Mr. O'Brien added that there is a weight-saving advantage in the low-tension equipment. The next weight is lb., a saving of about 12 lb. over a single high-tension unit, and nearly 120 lb. over the usual multiple unit.

Boeing Aircraft's T. F. Turekoff said his company had had wholly satisfactory results from temporary testing of the low-tension system installed on the Hercules engines powering the new Hercules IV transports for BOAC. The equipment successfully went through both the 120-hr test for certification and the 800-hr type test.

Moisture

Considerable discussion of the problem of moisture-condensation in the ignition "circuit" showed great differences of opinion as to whether or not this was an important source of ignition failure.

Representatives of BOAC reported that in the Far East route, where time flying hours have been operated well just recently, it was a common occurrence to drop an egg-size ball of water out of each electrical lead. G. B. Shaw, of BOAC's Far Eastern line, and he was convinced that this condensation was due to the coming down of the engine while in the water, in condensation at 100% humidity, and that the water did not find its way into the leads from any air intake. The advice thought this was probably their worst source of ignition trouble, since it caused a complete loss of spark.

But Shaw admitted that their records were unimpaired as to the actual cause of failures of the ignition system,



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inter, as the patient to get the engine running again, the carter's mechanics would commonly change the plugs first—as being the easiest thing to do and most likely to get the demand result. Hence, the agency wouldn't always reflect the real cause of the failure.

To keep the condensation out of the plug duct, BDAC has devised a rubber grommet to fit into the top of the screen tube.

To solve the condensation problem, BEAC has tried, with success, controlled oxidation of the 'barnes,' through a series of absorbers.

“It’s a firm seal as long as the water could be kept out of the barrel of the plug there was no need whatever to worry about condensation in the bar-
rel.”

On the other hand, Roda-Roda's West felt that heavy condensation could be regarded as a cause of plug-laden condensation could lead to muffling, this would cool down and plug as compared to the other resulting in the building up on that plug of lead deposit and eventually causing the plug to catch.

An Observer, A. F. Hutton, RAF, commented that harness construction is still a problem—citing especially the RAF's recent experience in Malaysia—but might be due in large part to the fact that the RAF was still using war time materials. He thought controlled problems offered the solution.

► **To Be Published**—Lodge Flags plans to make available a complete transcript of the two-day gathering as soon as the verbatim report has been edited.

Strain instead of air is being used in a model tunnel at Pratt Institute, Brooklyn, N. Y., to demonstrate the need for an expensive electric powerplant and compressor system.

Developed under the direction of the engineering school's Major James B. Randolph, the facility is connected to the Institute's 115 psi steam loop. Some of the test sections is 5 sq. in. inside throat area in 5.58 sq. in.

According to Riedel¹⁰, steam pressure is reduced to 5 psi at the mouth; the temperature value being about 205 F. Passing through the nozzle, pressure drops to about 1-2 psi, giving a temperature of 108 F. Speed at this point is equivalent to $M = 1.384$, it is observed.

No logging is reported, except at start-up.

One experiment contemplated is to use drums at a little below the saturation point to avoid the use of soldiers. This means, Randolph says, that the drums would be cloudy on one side of the shock wave, clear on the other side.

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FINANCIAL

CAB Asks Airline Securities Role

Agency seeks power to regulate carriers' financial structures; lines oppose this extension of control.

The Civil Aeronautics Board is again making a strong bid for jurisdiction over sources of airline securities. The Board contends that, for effective discharge of its duties, it should have power to regulate capital structures of airlines. The industry at large, on the other hand, is opposing this additional grant of power to the CAB.

The airlines, like all industry, resent all intrusions of governmental controls and what might be considered undue interference with private enterprise. But the carriers are endowed with a public interest and enjoy a franchise which provides them with a degree of monopoly. And financial assistance is afforded the industry through subsidy and payments and other forms of government support.

The need for security controls has been clearly stated and accepted in other regulated industries. The Interstate Commerce Commission and the Federal Power Commission have such power with respect to the public utilities under their respective jurisdictions. "Prudent Cost"—the criterion of rate controls in other regulated industries is cited regularly by the CAB as precedent for granting similar authority over airline finances. In its current attempt for such power, the Board maintains that certain airlines "failed to take advantage of the favorable equity market which occurred during and immediately following the war" and incurred their debt too heavily.

The airline position, frankly conceded by the financial community, is to the effect that the CAB is inadequately equipped to promote the public interest in problems of airline finance.

There can be no question that an over-all view of the regulatory process is essential to determine the place, if any, of jurisdiction over airline finances. Inquiries also arise as to whether the CAB could have developed sound capital structures among its airlines if it had power to pass over issuance of securities. Since any grant to one-duplicate of the current position and to the fact that the Board now have almost industry optimism in raising funds which today provide unequal competition and the security of any airline program.

Consequently, it is much that the CAB can do today in controlling

line capital structures without the grant of any additional powers. But such elements of control frequently appear to operate at cross-purposes.

A prime example is posed in the Board's relationship with the Reconstruction Finance Corp., in assigning government loans to air carriers. Under the law, the CAB is required to certify to the RFC that the applicant can discharge its obligations without benefit of a federal recapitalization before RFC can make the loan. This requirement has created a series of paradoxical situations.

In its recent requests for control over airline securities, CAB has persistently stated that it would have discouraged heavy debt structures but could have presented more equity issues in the past. Yet, the Board has not only asked the RFC to do so, but also refused to allow itself to be placed in a tight corner in the case of Northwest Airlines.

Look at NYAA—In this particular instance, there are a series of conflicts in policy. In an decision of April 29, 1949 in which Trans-Atlantic and World were made in Pan American and American Overseas Airlines, the Board issued its "black check" mandate. Believing that these carriers might be acquiring excess capacity in the Boeing 367-400s, the Board warned that "Because of the Civil Aeronautics Act is not a black check which airline management may fill in for any amount which it feels necessary to request without quantity or type of service which that management may see fit to provide."

A few months later, CAB questioned an RFC-guaranteed credit of \$21 million to Northwest so that the carrier could fill its obligations in acquiring 10 Stratliners. Yet the question of excess capacity is far more obvious on Northwest's current route than in the service of Pan American or AOA.

While the anomaly demands of the Board approving the RFC credit is noted that it was not understanding long-acting of the Northwest management concerning the aircraft involved, such philosophy may now be placed to a serious test.

Facing directly as the CAB-opposed Northwest capital structure (by

its certification to the RFC), it is apparent that the carrier's debt is the sole regulatory solution to its existing equity problem. About similar current actions pertaining to TWA's heavy debt structure led a CAB member to declare last year that if the Board had the power it would have told them "they had to have a different kind of capital structure."

It is noted that in its agency actions on the Northwest RFC-guaranteed credit, the Board has far been about that carrier's basic problems. If anything, it has compounded a series of errors which will continue to be troublesome until corrective action—which should have been taken in the first place—is finally achieved.

Time Lag—The Board's case for control over securities issued by the airlines may also suffer from time-consuming delays which have been present in its rate scale and mail rate proceedings to any satisfaction of airlines because of its great importance. Market fluctuations can develop almost overnight upsetting the best-considered financing program. It is well known proposals had to meet clearance by the CAB in the same manner in which rate matters are determined, the air carriers would find it extremely difficult to market securities to good advantage.

These delays in rate proceedings are costly in themselves and have an adverse bearing on the making of capital structures.

For example, in the pending litigation of a lawsuit of carriers are very much concerned as to the adverse decision pertaining to the Los Angeles-Honolulu route award. Large investment monies available on the part of the carriers involved must find financing in the meantime. This factor alone frequently serves to discourage carrier support of the airlines.

The question of jurisdiction over airline finance is clearly one phase of the overall industry problem. It is not yet enough to control stability in the group.

All efforts must first be devoted to develop consistent earning power. This will sell off for drastic rate reductions and the elimination of excess competition within the industry. Heavy subsidy payments do not add to an airline's credit standing. Earnings which are largely attributed to this source are very suspect among creditors. On the other hand, earnings power generated by a company's own efforts inevitably to create a much higher credit valuation.

With reduced and sustained earning power, as its carrier has much greater flexibility in arranging such financing in light with its needs—and it will make little difference in the final analysis which government agency has jurisdiction under such conditions.

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EQUIPMENT

Mounting Bracket Is Big Timesaver

Westinghouse's new "bottomhole" bracket saves an hour in generator mounting; other aircraft uses seen.

Up to an hour's saving in generator mounting time has been realized at Wright field by use of a new, "bottomhole" mounting bracket recently patented by the Westinghouse Electric Corp., East Pittsburgh, Pa.

The same principle may be applied to the growing number of engine-driven accessories, when they are readily accessible. The nuts are threaded on the studs far enough to leave just a fraction of an inch between the nut and the pad for the hinge portion of the bracket. After installation, the only operation remaining is to rotate the generator so that the nuts coincide with the specially provided, counterbored studs and tighten the bolts.

The rim and flange portions of the bracket are joined by longitudinal reinforcement ribs which provide strength and rigidity in the direction of the vibratory stresses to which the unit is subjected, while keeping the weight low. Ample space is provided to make the nuts accessible.



DRAWING of general Westinghouse generator mounting bracket showing slotted mounting attachment holes.

Since as many as nine accessories are mounted on today's aircraft engines, since that a whole new day could be an end of all accessories equipped with this unit, according to some estimates. This principle is a complete potential new way of saving during engine removal and build-up or accessory replacement, as well as reduced ground delays caused by accessory failure.

Wright-Patterson AFB claims a saving of 50-60 minutes in mounting A-1013 100-hp. generator on B-36's.

New T-28—Principal features of the bracket are the elongated, unslotted openings, one end enlarged to permit the hold-down nut to pass through. By allowing the nut to be worked on the engine pad, study prior to mounting the generator, a considerable shortening of the mounting bracket proper is accomplished. This saves a third of a pound.

Moves the generator center of gravity as close as possible to the mounting pad, resulting in a reduction in the susceptibility of vibrating stresses set up in the generator by engine vibration. Many generator failures have been attributed to excessive vibration.

Reduces overloading moment on the engine rear case and generator pad.

Provides access space to the congested engine accessory compartment.

The mounting bracket is derived from the fact that the nuts can be installed on the studs before mounting the generator, when they are readily accessible. The nuts are threaded on the studs far enough to leave just a fraction of an inch between the nut and the pad for the hinge portion of the bracket. After installation, the only operation remaining is to rotate the generator so that the nuts coincide with the specially provided, counterbored studs and tighten the bolts.

The rim and flange portions of the bracket are joined by longitudinal reinforcement ribs which provide strength and rigidity in the direction of the vibratory stresses to which the unit is subjected, while keeping the weight low. Ample space is provided to make the nuts accessible.

and indicator stem are polished hard chrome. Indicator stem is fitted with thread adapter which provides up to down adjustment of the indicator bulb two point to flat, convex or concave surfaces. Vertical adjustment of 1/8 in. can be made.

Total indicator weight is 0.075 lb., with a maximum production of 4005. Unit is designated Model No. 1-85M and weighs 7 1/2 oz.



Heating System For Jet Fuels

A system for heating hydrogen peroxide fuel tanks has been developed by A. V. Roe Canada, Ltd. This will prevent the chemical from freezing and ensure fast, reliable ground starting of jet engines.

Use of hydrogen peroxide for jet engine starting is advantageous because of its rapid decomposition into oxygen and steam upon exposure to a catalyst, and the heat thus produced in the process. Dowtherm is its heat transfer agent (11 deg. F.), a problem which A.V.R.'s heating system is specifically designed to overcome.

Operation—When the temperature of the hydrogen peroxide, contained in an upper reservoir, drops to a given value, a thermosensitive switch mounted on the fuel bottom opens a heating valve, allowing a small amount of the chemical to flow through a non return valve to a catalyst contained in the decomposer. The heat vapor generated in the decomposer passes first through a liquid trap then a heating coil heating the fuel up to desired temperature which activates the thermosensitive switch to close the heating valve and stop the flow of fuel. Excess vapor is condensed in a heated overflow.

The entire unit is compact and lightweight. Simplicity of operation should preclude any maintenance problems that might arise.

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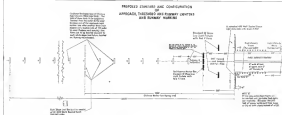
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CAA Pushes Approach Light Plan

But ALPA still prefers its own center-line system to the left-hand, single-row, ladder-type program.

The Civil Aeronautics Administration is stepping up the voltage on its lagging high-intensity approach light program. Between 9,000 and 15 of the lowest U. S. airports are slated to glow brighter by Nov. 1. But the Air Line Pilots Assn. is likely to make more spots fly before CAA's plans are carried out.

Present indications are that the 15 sites being studied by CAA will have left-hand, single-row, ladder-type, high-intensity approach light lines. The fed-aid agency believes this procedure will mainly suit users of the new lighting facilities and lead to quick installation.

Configuration Dispute—Wrangling over the proper configuration has already delayed the multi-authority-dollar approach light program. CAA feels that unless it starts installing the lights within the next few weeks they will not be ready for bad weather and winter.

ALPA says the left-hand configuration is undesirable and will make a determined fight to have its own center-line system installed. The pilots' union insists that the center line was not evaluated fully while undergoing CAA-sponsored tests at Aerob, Calif., and elsewhere.

The industry is the principal opponent of the center line. It claims to endorse a system that does not have a 100-ft clear over-run area. A jet fighter reportedly walked out part of a center-line system during a recent land-

ing at Indianapolis.

Besides, the military thinks the center line increases the danger of cockpit error due to search area stress. It is argued that watching a center line would be extremely difficult from some plane cockpits even at a near-down landing attitude.

Arrives in Middle—While type objectives of at least two major airlines including the ALPA viewpoint, most carriers feel they are holding the bag as the approach light squabble. They want improved lighting to permit increased maneuvering and greater schedule reliability. The configuration, they feel, is of minor importance compared with the need for quick action.

Airline operations officials are expected to make an important policy decision on the high-intensity approach light controversy before the end of this month. It is feared that ALPA pilots may refuse to use the slope line or left-hand, single-row configurations. If CAA won't budge from its position, ALPA might go to Congress to the detriment of the whole lighting program.

Program Outlook—These reports have been cited tentatively for high-intensity approach light installations by Nov. 1: Baltimore (Fleetwings), Philadelphia, New York (LaGuardia), Boston, Atlanta, Detroit (Willow Run), Houston, Dallas, Kansas City, St. Louis, Oakland, San Francisco, Bu-

ford, Seattle and Portland, Ore. Land acquisition difficulties at some fields may make it necessary to install lines of less than the desired 500 ft. Lack of a 700-ft line may mean construction of a project with a view to transferring it to another location. On the other hand, lines of less than optimum length may be installed now and lengthened later.

Where existing 1,500-ft., low-intensity, red center lines are involved, the high-intensity units will be located on the same left-hand line and stationing. The line already has been installed, and at Chicago, where a 1,500-ft line is in place.

Agreement on Features—There is no disagreement on the type of light that should be used. This type of light of the slope line but with the low-intensity lights of about 90,000 candela-per-foot. Existing red nose lights, installed at more than 50 airports, are said at ALPA to be satisfactory. They have been used for over 15 years.

More than a year ago, over ALPA protests, the U. S. Maritime Board approved the CAA-backed, double-row, forward-sloped slope line configurations as the U. S. standard. It was said that the slope line configuration gave the sharpest indication of direction, the only accurate indication of altitude among systems tested, and the most accurate information as to lateral position with respect to the approach axis.

During the past year, high-intensity approach lights with the slope line configuration have been announced at Washington National, Towson (New York) and Los Angeles Airports.



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Then, about five months ago, GAA withdrew its support for the slope line modification. (Aviation Week Jan. 23). Studies had shown that terrain problems made the slope line pattern impractical for many airports. As a result, the single bar, left-hand system is being considered as an alternative to the double bar.

► **Experiments Continue**—GAA is still experimenting as an effort to improve both the slope line and left-hand configurations. Rightly it considers the threshold and 1000 ft and 1800 ft out from the approach and all the runway have been added to the slope line system at Los Angeles Airport. Red lights can be placed over the white light on both the slope line and left-hand, single-line systems.

Use of red lights lowers the conspicuity of the standard white lights by 78-83 percent. Tests are also under way to cut down the number of lights in the slope line system, without the runway, or to reduce the intensity of the lights closest to the threshold.

Slope line fixtures now being installed by GAA were brought with them in 1949. High-intensity lighting fixtures in the fiscal 1950 budget were devoted elsewhere.

► **Progress Cast**—The fiscal 1951 budget originally called for 30 more installations of the slope line configuration, costing an average of \$107,000 apiece. But the House reduced the program to 13 installations costing \$1,501,735.

Use of the approach, left-hand system, left-hand, and single-bar approach light system over other projects, since fewer lights and false-emancipation less have are required. However, the cost of the power supply and control wires link between single- and double-line configurations.

In explaining its opposition to both the double-line slope line and single-bar, left-hand configuration, ALPA says that studies of these patterns is performed by pilots who participated in light evaluation tests.

These weaknesses are attributed to ALPA to the dual row slope line. It is unable to determine left from right row when both rows are not observed simultaneously. (This is particularly noticeable during low daytime visibility.)

► **Confusion between approach and runway lights, resulting in landings in the approach area.**
► **Color differentiation of the rows is impractical.**

► **Two rows which converge toward the runway create a false optical illusion at staggered glideside in pitch angle.** (Two cases the runway is apparent.)

► **Poor color-line identification under low visibility conditions.** Natural light tendency is to hug the row of lights

if the pilot guides on the right-hand row, resulting in for the left, a missed approach is likely to result.

► **Cost of acquiring land is too high.**
Left-hand row tends, according to ALPA, as:

- **No centerline identification.**
- **Confusion when approaching in a cross wind.** (The plane appears to be either closing in, or drifting from, the lights, with a resulting tendency for the pilot to change heading, in order to parallel the lights. If this does, the plane will drift off course.)
- **Visible only to pilot in the left hand seat.** This reduces the safety factor of the second pilot.

The pilot group adds that the slope line and left-hand systems do not increase the importance of defining the end of the runway as possible for guidance beyond that point.

► **ALPA system**—Center line motion detector system is more effective than the slope line. It is used to eliminate the possibility of confusion, to give positive and early identification of the runway, and strong definition of the runway center line and its effect on the runway back gradient 1000 ft out and again at the runway threshold.

Lights in the ALPA center line system are the standard 14 ft, slope line has been continuing 16 ft, standard, 400-watt lamps. The new system 3000 ft from the runway center line, with one color bar 1000 ft out.

► **Super-Brightness**—The condenser discharge type of lighting, high intensity white light, and red light, approach light bar used. These special lights have extremely high power-up to several billion condenser—and flash from the center to the most end of the approach light system 125 feet in a minute.

ALPA says the distance of the flash is so short—17 to 200 seconds—so that it does not affect day or night vision. The flash is described as appearing like giant tower lights fired toward the runway. It is used to provide a "point source" of light against the haze into glow from steady burning lights during rain, snow or fog.

The condenser discharge lights are of great value in identifying the center of the light row and ensuring the pilot that the inner bar will soon be visible, according to ALPA. These flashing units can be seen when at an angle to the runway, as when circling, and stand out against other city lights. Disposed characteristics of present solid-beam lights make these visible only when the plane is lined up with the runway.

► **Used were 1947**—Original condenser discharge lights were installed at airports in 1947. They were installed at Ansett, and Cleveland has a partial system. ALPA believes they will make up for the loss of intensity of red lights are used on the slope line bars.

Centers of the condenser discharge lights are no less expensive and are designed to give a flash. Cost of one type of condenser discharge unit is estimated at about \$1500 per unit and another type at about \$2500 per unit, compared with the comparatively inexpensive \$115 per slope line unit.

The ALPA center line system also has a 200-ft clear over the area and level as and aligned lights and green lateral threshold lights. A double row of approach side steps is painted along the runway center line to insure ease guidance.

Mail Pay Rise Will Put CAL in Black

Continental Air Lines will soon rub out the red ink tentatively placed on its books during 1949, thereby making the year profitable for month all the its domestic trunk operation.

Initial figures show Continental with a \$270,000 net loss for 1949, making it the only domestic trunkline, besides Northwest to finish in the red. But now, the Civil Aeronautics Board has offered to increase CAL's 1949 and pay by \$477,000, giving the company a substantial profit for the year. Mail rates in 1950 will also be raised.

► **Wedge Back**—When CAL's action wasn't all to the good. The Board decided that the carrier's \$183,000 net profit in 1949 provided more than a 7 percent return on company investment. It said pay for that year was about \$61,000 and leaving CAL with a 14% profit.

In analyzing Continental's deficit operations last year, CAL found that the sharp drop in passenger load factor from

49 percent in 1948 to 49 percent in 1949 could be traced to the company's reduction of 15 passenger DC-3s to 40 passenger Constellation. Passenger miles flown, gained more than 11 percent during the year, but not enough to fill the 14 percent seats now available.

Nevertheless, CAL found no fault with Continental's decision to buy new Constellation. When the order was placed in 1948, it seemed that DC-3s would have to be disposed of by the end of 1948 because they were unable to meet CAL's transport category requirements. (The decision has since been extended more than two years to Dec. 31, 1951.)

► **Fixed Prices**—Reuter, CAL contracted for its Constellation at a price considerably under its present quotation. The Board points out that Continental will need a DC-3 replacement in the next several years. It feels there is no reasonable question whether Continental could ever get new transports at a price equal to, or lower than, the 1946 price of the Constellation.

In view of the low fixed density in CAL's books, it is natural that economic mitigation of Constellation on the carrier's system has proved difficult CAL said. However, the Board found that Constellation had not shown a tendency to over schedule and as traffic grew in future years the Constellation would begin to pay off.

Continental's future mail rate in the incentive sliding scale, variety of service aircrafts decrease in passenger load factor rate. CAB thinks the new rates will make CAL to break even at a 37 percent passenger load factor and make it 6 percent profit at the expected 45 percent load factor.



STANDING ROOM ONLY

Completion late last year of a \$15.5-million extension of Washington National Airport's administration terminal building will not, not eliminate the housing shortage at the nation's third-busiest commercial field. Clark American Airlines Administration, which operates the airport for the government, says available terminals have already applied for more space than is available in the new

structure. A dog house and hen house appear to be two new additions to the area of accommodations now in the terminal. Transporting airline officers being the air and south command of the present building will move into permanent quarters, permit expansion of working room space in the lobby. Extension adds 11,000 sq ft of space to the terminal's present 115,000 sq ft.



This year, we at Edo are entering our 25th Anniversary. On September 28, 1950, we will have completed a quarter of a century of service to the aviation industry, the service industry, and, more recently, the field of electronics. During these years, Edo Flying Fish has been a familiar mark on a wide variety of projects ranging from airplane floats to intricate electronic underwater detection equipment.



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The "Mako" project led to the development of Edo airplane floats which have since earned a world-wide reputation for performance and service. The existing skills required in building light, strong, leak-proof floats have since been applied in the design and manufacture of many different electronic aluminum components. And now our Electronics Division is planning a leading role in the development of significant underwater detection equipment.

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MANSFIELD—Turbo-prop in traffic snarl.

BEA's Choice

Mansfield sees needs for piston-engine, turbo-prop and jet transports.

The chief executive of British European Airways, which next year may become the first carrier in the world to fly turbo-prop transports in scheduled service, says tests have already determined the proper niche for his own ships alone competing with turbojet and jet-propelled aircraft.

Peter Mansfield believes that for some time to come turbo-prop transports may be the most desirable one available up to 400 miles. "For stages of 500-1000 miles, the jet-prop is once more very attractive," he declared, adding that "for stages of 1500 miles or over the straight jet is the answer."

Travis Mansfield-Mansfield, former director of Imperial planning and development for Britain's Ministry of Civil Aviation, later his chairman on operational operations with the de Havilland Comet jet transport and the turbo-prop Vickers Viscounts. These two aircraft, Mansfield asserts, have set a new standard of passenger comfort which will draw traffic to the lines using them.

BOAC has ordered 14 Comets and BEA has ordered 20 Viscounts. The 30 passenger Comets are expected to go into commercial service in Europe before early in 1971, and the 40 passenger Viscounts will be placed on Continental routes soon afterward.

Easy Maintenance—Mansfield said after 3000 miles of flying around Europe, the only maintenance operation performed on the Viscount was the changing of an oil pump actuator and the topping up of oil in one leg. The Comet, he declared, has already flown more than 200 hours and also is proving

that lack of vibration in turbo-prop aircraft means low maintenance costs.

Comparing the three types of transports on a 300-mile route, says Peter Mansfield, Mansfield estimated the turbo-prop ships would operate at \$2.00 per mile, the turbojet at \$2.22 and the straight jet at \$2.50. This compares six 4-passenger planes of each type flying seven hours a day.

Cost Comparison—On an hourly basis, the operating cost of \$120.50 for the piston type, \$141 for the turbo-prop and \$221.22 for the straight jet shows a much larger spread. But since the turbo-prop planes would make the same number of trips in fewer flying hours the total costs are comparatively close.

When on a 300-mile run the turbojet would be about 20 percent more expensive than a piston transport, but the situation changes as the route lengthens. On a 900-mile stage, Mansfield estimated, plane-mile operating costs would be about the same for the piston transport and the turbo-prop, with the jet at 11 percent higher. In view of the added passenger comfort for the turbo-prop equipped with the turbo-prop engine, this, he says, is not a bad bargain.

Over the 1800-mile stage, the turbo-prop shows a one-hour time gain and the straight jet 2 1/2 hr over the piston type. Operating costs are again about equal for the piston and turbo-prop types, but the fast turbojet, with costs now only 9 percent more, would be likely to attract dominance.

Another Late 631 Flying Boat Last

(McGraw-Hill World News)

Form-France's 70-ton, six-engine flying boat Latécoere 631 became a strong contender for the title of the world's smallest airplane last month when one of the ships planged into the sea off the French west coast recently taking its last ever test of 10.

That was the latest of a series of crashes which has whittled the original fleet of eight boats down to four. It probably will be the last ever odd as they now will be grounded for good.

Crash of last month's crash still is unknown. The last was on a test flight when it exploded and the wreckage pitched into the sea.

Tragic Ship—The Latécoere's test was as dramatic as it was tragic. The ship was designed and two prototypes partially built before the war. The Germans captured one prototype, finished it and flew it to Lake Constance where it was sunk by the RAF. The second prototype was disassembled by the French Resistance under the nose of the Germans and hidden throughout the war.

Right Later was built for Air France after the liberation. The first one three two passengers while flying up the coast of South America, before the pilots who had been invited on the flight to write up the plane. The Argentines abandoned plans to buy a fleet of Latécs.

The Latécoere test was put on Air France's schedule run. Despite the fact that they were very comfortable and carried 30 passengers they flew in the end due to high operating cost. Only a few months later a 631 pitched into the English Channel in a snowstorm. The weather and inadequate navigational and communication equipment were blamed.

Thus in July 1948 a Latécoere came from the Andes on a scheduled Air France run disappeared without a trace with 55 passengers and crew aboard off the Andes. Air France promptly initiated its search of the flying boat.

A commercial company was formed to redesign and perfect the remaining 631s with the object of using them eventually in scheduled cargo planes. A successful last flight from Reims to Bordeaux, Frenchland, never had been completed by one of the ships just before last month's accident. Chances are it will be the last.

Western Plans to Dissolve Inland

The nation's 16 domestic transients now may be reduced to 15.

Western Air Lines in planning complete dissolution of Inland Air Lines, now operated as a division of WAL, Inland has not been an independent carrier since May, 1948, when the Civil Aeronautics Board approved Western's acquisition of control through purchase of 33 percent of the smaller carrier's stock. But Inland has continued to act as a traffic and revenues separately from Western.

Now, however, Western has decided against continued existence of Inland as a separate entity. Steps to dissolve Inland, pursuant to Wyoming law, will be initiated in six months or less, if approved. Since 1948 W.A.L.'s stock in Inland has grown to nearly 90 percent.

SHORTLINES

► Air Line Dispatchers Association—Reports a new contract with Boeing providing a starting wage of \$375 a month, the scale rising to \$550 a month in the seventh year. Foreign strike strikes a \$180 per month. Plans to be effective to July 1, 1971.



Mansfield of the lightweight efficiency and trouble-free performance of Hartman's current customers and other d-c devices in military and civil aircraft. Jack & Heintz called on Hartman to supply vital

relays for the J48 GC-18 control panel installed in the Stratofort. Each of the aircraft's six generators is protected and regulated by an individual GC-18 control panel equipped with five Hartman relays.

- (1) Differential Voltage and Reverse-Current Relay—Controls generator to bus when generator voltage exceeds battery voltage, disconnects generator from bus upon reversal of current.
- (2) Ground Fault Relay—Senses ground fault, where fault exceeds set value, opens to interrupt generator.
- (3) Overvoltage Inhibitor Relay—Prevents load current to direct generator protection overvoltage and automatically acts on overvoltage relay to trip or lower voltage thus after five relays.
- (4) Equalizing Relay—Diagnoses regulator equalizing circuit from regulator bus to avoid pulling system voltage down when generator is inoperative.
- (5) Overvoltage Relay—Senses overvoltage and cuts off generator. Relay has inverse time characteristics to prevent nuisance trips.
- (6) Converter and Dropout Relay (Not shown)—Located in fuselage near main bus, one of three protection units, each consisting of a GC-18 panel, converts generator protection from bus voltage to bus voltage and generator protection.

Typical of Hartman design and manufacturing, relays in the B-47 are just a few of the many d-c devices engineered for the aircraft industry. Whenever your problem involves d-c controls, turn it over to Hartman.

where it will receive prompt attention . . . where it will be analyzed and engineered with an efficiency that comes from nearly half a century of specialization.

the Hartman Electrical Mfg. Co.
MANSFIELD, OHIO

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Announcing . . .

The Aviation Week Fellowship In Aeronautical Engineering

Human progress is based on knowledge. A time-honored test of the standing of an individual (or organization) in a community is the support he gives to the educational system. By that test, the aeronautical industry has been remiss in obligations stemming from its citizenship in the aviation community.

Aviation Week now is completing a national survey of scholarships and fellowships in aeronautical engineering awarded by individuals or firms in aviation. Results of the survey will appear in a forthcoming issue, with a complete list of all accredited universities and colleges offering approved courses in aeronautical engineering.

But already it is obvious that the aviation industry—so dependent for its progress on research and advanced study—has done and is doing all too little to promote such study.

For a student of modest means, advanced study may be impossible because of its cost. Awards of awards, prizes, tips and other such honors to a student only signify a certain recognition of his work, but the most practical assistance he can use is financial aid to complete his studies.

In appreciation of its role in the aviation community, Aviation Week is pleased to be able to announce the establishment of the Massachusetts Institute of Technology of the "Aviation Week Fellowship in Aeronautical Engineering."

The Fellowship carries a stipend of \$3000 per year, and is awarded annually to a male U. S. citizen enrolled for an advanced degree in the Department of Aeronautical Engineering of MIT.

The recipient is selected by MIT according to its own policies. This publication does not define the requirements nor choose the winner.

The progress of aeronautical has vastly increased in complexity. It has been stated that the airplane is the most completely engineered man-made structure, reflecting facets of every science. Until recent years, metallurgy, chemistry, electronics, aerodynamics and some other sciences were vital to the aeronautical engineer.

From these sciences the aviation industry now is drawing the strength of progress. Aviation has sought to gain from such progress. Only a dynamic industry survives. Yet, it can be shown that the aviation business is falling behind in providing the necessary tool-knowledge—that makes progress possible.

Greater knowledge means more study, the expense of which often deters promising students. A noted educator recently pointed out that many of the most promising students of our colleges and secondary schools were those who would be hampered by being required to pursue advanced study. Aviation looks to the laboratory and the classroom for its future growth. Is it not fair to suggest that aviation should aid those students?

Aviation Week believes it is part of the aviation community. It, too, will progress only as the aviation progresses. It has long sought new ways to contribute tangibly to that progress. The Aviation Week Fellowship is a part of this program.

Winner of the Fellowship for the academic year 1950-51 is Herbert Matthias Voss. Mr. Voss is in the Heisen Group in MIT's Department of Aeronautical Engineering at MIT and will receive both his Bachelor of Science and Master of Science degrees in June, 1951. Mr. Voss was selected by the Department of Aeronautical Engineering at MIT, with the approval of the Committee on Graduate School Policy.

It is intended that the Aviation Week Fellowship be awarded annually to a recipient who qualifies. In this manner, an earnest student of the sciences of aeronautics may carry on his work for several years under the Fellowship award.

Leadership always brings responsibilities. Aviation Week, as the leading U. S. aeronautical publication, a handful of its responsibilities to the aviation industry that made its success possible. It is in this sense that Aviation Week is gratified to be able to offer its services in a citizen of the aviation community in furthering the pursuit of aeronautical knowledge.

We are not jealous of this role. To the contrary, we hope that others in aviation will give attention to the extreme necessity of aiding aeronautical study. It would be beneficial to aviation if future surveys of aeronautical engineering scholarships and fellowships could include the names of many more aviation leaders.

AIRHEAD

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Exercise Swamper, the all-weather rescue in North Carolina closed in May on a high note of success. It proved that an entire airlift can be established, supplied, and supported entirely by air.

Contributing much to the success of "Swamper" was the performance of Fairchild C-119 and C-82 Packers.

The new C-119s proved their usual loads with flying colors—and well they might, because this was a modern airlift for the Fairchild planes, with quick and easy loading and unloading of men, equipment and other loads, supplies.

Packers, specially engineered and built for use by America's aerial air and ground forces, are proving themselves every day, under all operating conditions.

FAIRCHILD AIRCRAFT CORPORATION
FAIRCHILD Aircraft Division



... of air pours through this intake when the new North American F-95 is flying top speed. Yet the J47 turbojet inside handles this easily and operates reliably, efficiently, and without vibration.

The Air Force's newest interceptor, a stablemate of the speed record-holding North American F-86, is designed for the high speed, high-altitude flight necessary to knock down enemy fighters and bombers. Teamed together, the F-86 and F-95 provide both offensive and defensive air power. Both use General Electric J47 turbojets for high performance under tough conditions.

As the G-E TG-190, this same engine has been certified by the CAA as the first axial-flow turbojet suitable for commercial use. In tomorrow's commercial transports, the TG-190 can provide the same speed, comfort, and dependability that are today built into the fastest and most powerful Air Force planes.

And in addition to the powerplant, General Electric also provides integrated engineering service that assures you of co-ordinated propulsion and electrical systems. From the designer's drawing board to the far-flung outposts of operational aircraft, General Electric's aviation experts can help you. Call your nearest G-E sales representative or write Apparatus Department, General Electric Company, Schenectady 5, New York.



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